## MECA0525 VEHICLE DYNAMICS

#### Pierre Duysinx

Research Center in Sustainable Automotive Technologies of University of Liege Academic Year 2021-2022

# Lesson 3: Dynamic vehicle stability

#### References

- G. Sander « Véhicules Automobiles», Lecture notes, 1983, Université de Liège
- G. Genta. « Motor Vehicle Dynamics: Modeling and Simulation ». World Scientific. 1997.
- J.R. Ellis. Vehicle Dynamics. London Business Book Limited. 1969

## Outline

- Bicycle or single track model
- Equations of dynamic behaviour of the single track model
  - Equilibrium equations
  - Compatibility equations
  - Differential equations of vehicle dynamics
  - Stability derivatives
  - Canonical form of equations
- Investigation of the vehicle dynamics stability
  - Sign of real parts
  - Investigation of the discriminant
- Steady state particular case
- Trajectory description

### Single track model

### Single track model

- Stiff vehicle
  - Pitch motion (q=0)
  - Pumping motion (w=0)
- No body roll : p=0
- One can neglect any lateral load transfer leading to a reduction of the lateral cornering stiffness when lateral accelerations remain below 0.5 g (L. Segel, *Theoretical Prediction and Experimental Substantiation of the Response of Automobile Steering Control*, Cornell Aer. Lab. Buffalo. NY.)
- Constant speed forward motion: V
- Symmetry plane y=0:  $J_{yx} = 0$  and  $J_{yz} = 0$

### Single track model

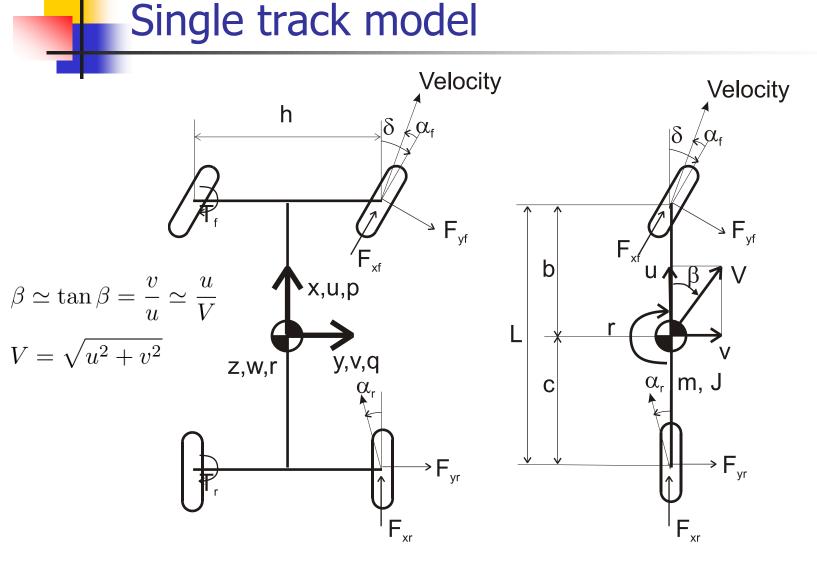
- Small angles and perturbations
  - Small steering angles (at wheel)  $\delta \ll 1$
  - Small side slip angles
  - $\Rightarrow$  Linearized theory

 $\sin \delta \simeq \delta \qquad \cos \delta \simeq 1 \\
 \sin \alpha \simeq \alpha \qquad \cos \alpha \simeq 1$ 

 $\alpha_i \ll 1$ 

#### CONCLUSION

- Linearized model with two degrees of freedom:
  - Body side slip angle  $\beta$  (v)
  - Yaw velocity r



Newton-Euler dynamic equations

$$\sum \overrightarrow{F} = \frac{d}{dt} (m \overrightarrow{V})$$
$$\sum \overrightarrow{T} = \frac{d}{dt} (J \overrightarrow{\omega})$$

Time differentiation in non inertial frame

$$\frac{d}{dt}\overrightarrow{V}\Big|_{absolu} = \frac{d}{dt}\overrightarrow{V}\Big|_{relatif} + \overrightarrow{\omega} \times \overrightarrow{V}$$

Equilibrium equations

$$\sum \vec{F} = m \frac{d}{dt} \vec{V} + m \vec{\omega} \times \vec{V}$$
$$\sum \vec{T} = \frac{d}{dt} (J \vec{\omega}) + \vec{\omega} \times (J \vec{\omega})$$

#### Dynamics equations of the vehicle motion

Model with 2 dof β & r

$$\vec{V} = [u \ v \ 0]^T \qquad \vec{\omega} = [0 \ 0 \ r]^T$$

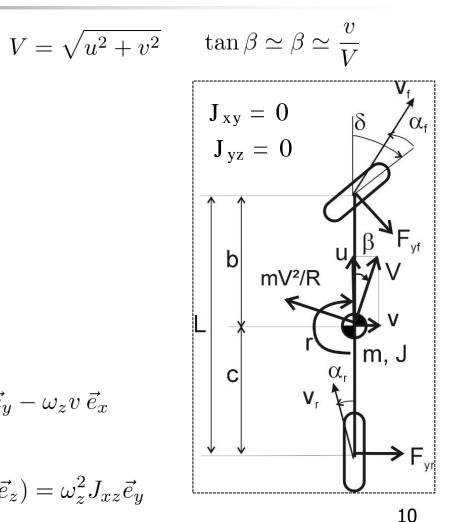
$$\vec{V} = u\vec{e}_x + v\vec{e}_y \qquad \vec{\omega} = \omega_z\vec{e}_z$$

Inertia tensor

$$J = \begin{pmatrix} J_{xx} & 0 & J_{xz} \\ 0 & J_{yy} & 0 \\ J_{xz} & 0 & J_{zz} \end{pmatrix}$$

It comes

$$\vec{\omega} \times \vec{v} = \omega_z \vec{e}_z \times (u\vec{e}_x + v\vec{e}_y) = \omega_z u \,\vec{e}_y - \omega_z v \,\vec{e}_x$$
$$J\vec{\omega} = [J_{xy}\omega_z \ 0 \ J_{zz}\omega_z]^T$$
$$\vec{\omega} \times J\vec{\omega} = \omega_z \vec{e}_z \times (\omega_z J_{xz}\vec{e}_x + \omega_z J_{zz}\vec{e}_z) = \omega_z^2 J_{xz}\vec{e}_y$$



### Dynamics equations of the vehicle motion

• Finally, dynamics equations write

$$X = \sum F_x = m (\dot{u} - \omega_z v) \qquad L = \sum M_x = 0$$
$$Y = \sum F_y = m (\dot{v} + \omega_z u) \qquad M = \sum M_y = J_{xy} \omega_z^2$$
$$Z = \sum F_z = 0 \qquad \qquad N = \sum M_z = J_{zz} \dot{\omega}_z$$

The only nontrivial equations are

$$Y = \sum F_y = m \left( \dot{v} + \omega_z u \right)$$
$$N = \sum M_z = J_{zz} \dot{\omega}_z$$

### Equation of motion

2 dof model

$$\vec{\not} = \begin{bmatrix} u \ v \ 0 \end{bmatrix}^T$$
$$\vec{\not} = \begin{bmatrix} u \ v \ 0 \end{bmatrix}^T$$
$$\vec{\not} = \begin{bmatrix} 0 \ 0 \ r \end{bmatrix}^T$$

Dynamic equations of motion

$$F_y = m(\dot{v} + ru)$$
$$N = J_{zz}\dot{r}$$

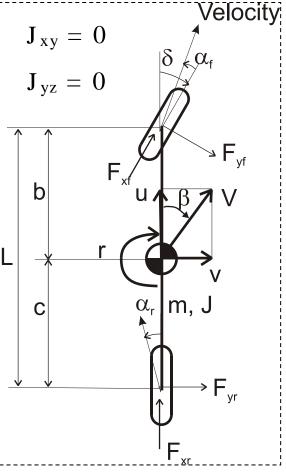
- Equations related to the fixed dof
- $\rightarrow$  Reaction forces / moment

$$F_x = -m r v$$

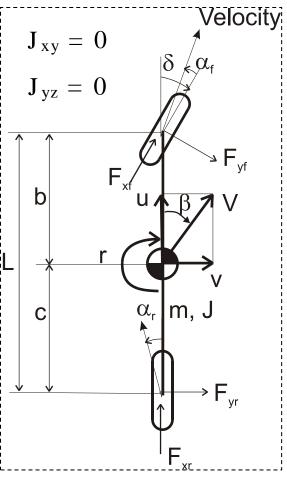
$$F_z = 0$$

$$L = J_{xz} \dot{r}$$

$$M = J_{xz} r^2$$



- Explanation  $F_x = -m r v$ Circular motion  $r = \frac{V}{R}$   $F_x = -m \frac{V^2}{R} \sin \beta$  $= -m \frac{V}{R} V \sin \beta = -m r v$
- Major working forces:
  - Tyre forces
  - Other forces (ex aerodynamic forces)
  - $\rightarrow$  Neglected because they don't depend on perturbations (in a first approximation)



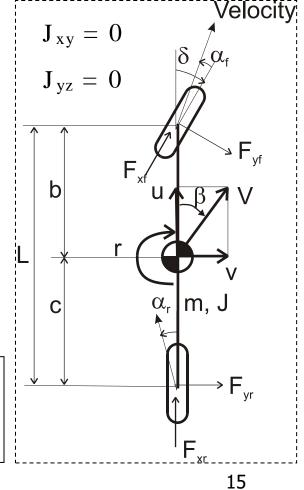
Equilibrium along F<sub>y</sub> and M<sub>z</sub>

 $m(\dot{v} + ru) = F_{yr} + F_{xf} \sin \delta + F_{yf} \cos \delta$  $J_{zz}\dot{r} = -F_{yr} c + F_{xf} \sin \delta b + F_{yf} \cos \delta b + T_{zf} + T_{zr}$ 

• Small angles assumption  $\beta \in [0^{\circ}, 15^{\circ}] \quad \beta \simeq v/u \simeq v/V \quad u = V \cos \beta \simeq V$  $\sin \delta \simeq \delta \quad \cos \delta \simeq 1 \quad v = V \sin \beta \simeq V \beta$ 

#### Linearized equilibrium

$$mV(\dot{\beta} + r) = F_{yr} + F_{yf} + F_{xf} \delta$$
$$J_{zz}\dot{r} = -F_{yr} c + F_{yf} b + F_{xf} \delta b + T_{zf} + T_{zr}$$



 If we neglect the self aligning torques and the tractive forces in a first step

$$\begin{cases} mV(\dot{\beta}+r) &= F_{yr} + F_{yf} \\ J_{zz}\dot{r} &= -F_{yr} c + F_{yf} b \end{cases}$$

#### **Compatibility equations**

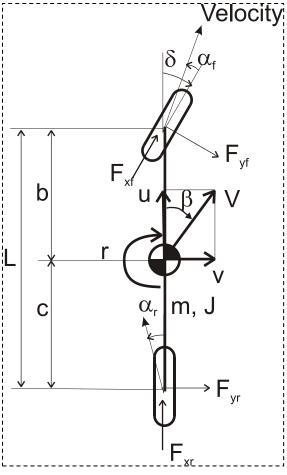
 Compatibility = relations between velocities and angles

$$\tan(\delta - \alpha_f) = \frac{br + v}{u} \qquad \tan \alpha_r = \frac{cr - v}{u}$$

Small side slip and steering angles

$$\alpha_f \simeq \delta - \frac{br+v}{u} \qquad \beta \simeq v/u$$
  
 $\alpha_r \simeq \frac{cr-v}{u}$ 

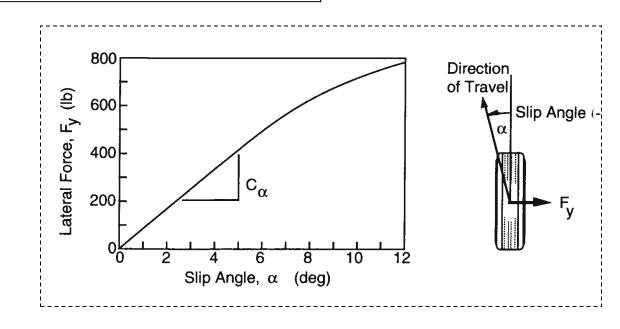
• If u=V 
$$\alpha_f \simeq \delta - \frac{br}{V} - \beta$$
  
 $\alpha_r \simeq \frac{cr}{V} - \beta$ 



#### Behavioural equations of tyres

Cornering forces and side slip angles

$$F_{yf} = C_{\alpha f} \alpha_f \qquad F_{yr} = C_{\alpha r} \alpha_r$$



Source: Gillespie (fig 6.2)

### Vehicle dynamic model

Dynamic equilibrium

$$mV(\dot{\beta} + r) = F_{yr} + F_{yf}$$
$$J_{zz}\dot{r} = -F_{yr} c + F_{yf} b$$

Let's introduce the behaviour law of tyres

$$mV(\dot{\beta} + r) = C_{\alpha r}\alpha_r + C_{\alpha f}\alpha_f$$
$$J_{zz}\dot{r} = -C_{\alpha r}\alpha_r c + C_{\alpha f}\alpha_f b$$

And then the compatibility equations

$$mV(\dot{\beta}+r) = C_{\alpha r}(\frac{cr}{V}-\beta) + C_{\alpha f}(\delta - \frac{br}{V}-\beta)$$
$$J_{zz}\dot{r} = -C_{\alpha r}(\frac{cr}{V}-\beta)c + C_{\alpha f}(\delta - \frac{br}{V}-\beta)b$$

### Vehicle dynamic model

 Reshuffling the terms in β, r, and δ, one gets the equations related to the lateral forces and the moments about vertical axis

$$mV(\dot{\beta}+r) = -(C_{\alpha f}+C_{\alpha r})\beta - (bC_{\alpha f}-cC_{\alpha r})\frac{1}{V}r + C_{\alpha f}\delta$$
$$J_{zz}\dot{r} = -(bC_{\alpha f}-cC_{\alpha r})\beta - (b^{2}C_{\alpha f}+c^{2}C_{\alpha r})\frac{1}{V}r + bC_{\alpha f}\delta$$

And so

$$mV(\dot{\beta} + r) + (C_{\alpha f} + C_{\alpha r}) \beta + (b C_{\alpha f} - c C_{\alpha r}) \frac{1}{V} r = C_{\alpha f} \delta$$

$$J_{zz}\dot{r} + (b C_{\alpha f} - c C_{\alpha r}) \beta + (b^{2} C_{\alpha f} + c^{2} C_{\alpha r}) \frac{1}{V} r = b C_{\alpha f} \delta$$

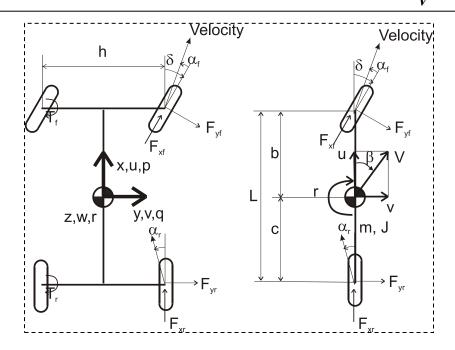
$$\int \text{Differential terms} \text{Terms in r and } \beta$$

$$Control \text{ terms in } \delta$$

$$20$$

#### Vehicle dynamic model

• Dynamic equations ruling the motion of the single track vehicle  $mV(\dot{\beta}+r) + (C_{\alpha f} + C_{\alpha r}) \beta + (b C_{\alpha f} - c C_{\alpha r}) \frac{1}{V} r = C_{\alpha f} \delta$   $J_{zz}\dot{r} + (b C_{\alpha f} - c C_{\alpha r}) \beta + (b^2 C_{\alpha f} + c^2 C_{\alpha r}) \frac{1}{V} r = b C_{\alpha f} \delta$ 



#### Stability derivatives

 Alternatively, it is the equivalent to <u>perform a linear Taylor</u> <u>expansion of the forces and moments around the current</u> <u>configuration</u> (that is reference configuration)

$$F_y = \frac{\partial F_y}{\partial \beta}\beta + \frac{\partial F_y}{\partial r}r + \frac{\partial F_y}{\partial \delta}\delta \dots$$
$$N = \frac{\partial N}{\partial \beta}\beta + \frac{\partial N}{\partial r}r + \frac{\partial N}{\partial \delta}\delta \dots$$

It is usual to denote them as *stability derivatives*

$$Y_{\beta} = \frac{\partial F_{y}}{\partial \beta} \qquad Y_{r} = \frac{\partial F_{y}}{\partial r} \qquad Y_{\delta} = \frac{\partial F_{y}}{\partial \delta}$$
$$N_{\beta} = \frac{\partial N}{\partial \beta} \qquad N_{r} = \frac{\partial N}{\partial r} \qquad N_{\delta} = \frac{\partial N}{\partial \delta}$$

#### Stability derivatives

Comparing with the initial developments,

$$mV(\dot{\beta}+r) = -(C_{\alpha f}+C_{\alpha r})\beta - (bC_{\alpha f}-cC_{\alpha r})\frac{1}{V}r + C_{\alpha f}\delta$$
$$J_{zz}\dot{r} = -(bC_{\alpha f}-cC_{\alpha r})\beta - (b^{2}C_{\alpha f}+c^{2}C_{\alpha r})\frac{1}{V}r + bC_{\alpha f}\delta$$

one finds the expression of the stability derivatives

$$Y_{\beta} = -(C_{\alpha f} + C_{\alpha r}) \quad (<0) \qquad N_{\beta} = -(b C_{\alpha f} - c C_{\alpha r})$$
  

$$Y_{r} = -(b C_{\alpha f} - c C_{\alpha r}) \frac{1}{V} \qquad N_{r} = -(b^{2} C_{\alpha f} + c^{2} C_{\alpha r}) \frac{1}{V} \quad (<0)$$
  

$$Y_{\delta} = C_{\alpha f} \quad (>0) \qquad N_{\delta} = b C_{\alpha f} \quad (>0)$$

### Stability derivatives

The equilibrium equations writes

$$Y_{\beta} = -(C_{\alpha f} + C_{\alpha r}) \quad (<0) \qquad N_{\beta} = -(b C_{\alpha f} - c C_{\alpha r})$$
$$Y_{r} = -(b C_{\alpha f} - c C_{\alpha r}) \frac{1}{V} \qquad N_{r} = -(b^{2} C_{\alpha f} + c^{2} C_{\alpha r}) \frac{1}{V} \quad (<0)$$
$$Y_{\delta} = C_{\alpha f} \quad (>0) \qquad N_{\delta} = b C_{\alpha f} \quad (>0)$$

Reorganizing the terms, one has

$$mV\dot{\beta} = Y_{\beta}\beta + (Y_r - mV)r + Y_{\delta}\delta$$
$$J_{zz}\dot{r} = N_{\beta}\beta + N_rr + N_{\delta}\delta$$

### Canonical form of the equations

 It is also valuable to notice that the single track model lag to a linear time invariant (LTI) model. It is usual to cast this model under the standard form

$$\dot{z} = A z + B u$$

The system state variables and the command vector are:

$$\mathbf{z} = \begin{pmatrix} \beta \\ r \end{pmatrix} \qquad \qquad \mathbf{u} = (\delta)$$

The system matrices A and B are easily identified and write

$$\mathbf{A} = \begin{bmatrix} \frac{Y_{\beta}}{mV} & \frac{Y_r}{mV} - 1 \\ \\ \frac{N_{\beta}}{J_{zz}} & \frac{N_r}{J_{zz}} \end{bmatrix} \qquad \qquad \mathbf{B} = \begin{bmatrix} \frac{Y_{\delta}}{mV} \\ \\ \frac{Y_{\beta}}{J_{zz}} \end{bmatrix}$$

### Stability analysis

Use <u>Laplace transform</u>

$$\begin{array}{ll} \beta(t) \to \beta(s) & r(t) \to r(s) & \delta(t) \to \delta(s) \\ \dot{\beta(t)} \to s \ \beta(s) & \dot{r(t)} \to s \ r(s) \end{array}$$

The system becomes

$$(s mV - Y_{\beta}) \beta(s) + (mV - Y_r) r(s) = Y_{\delta} \delta(s)$$
  
-N\_{\beta} \beta(s) + (s J\_{zz} - N\_r) r(s) = N\_{\delta} \delta(s)

 The stability of the free response stems from the study of the roots of the characteristic equation

$$\Delta = (smV - Y_{\beta}) (sJ_{zz} - N_r) + (mV - Y_r) N_{\beta} = 0$$

 $mVJ_{zz} \ s^2 - (Y_{\beta}J_{zz} + mVN_r) \ s + (Y_{\beta}N_r - Y_rN_{\beta} + N_{\beta}mV) = 0$ 

Characteristic equation 

$$mVJ_{zz} \ s^2 - (Y_\beta J_{zz} + mVN_r) \ s + (Y_\beta N_r - Y_r N_\beta + N_\beta mV) = 0$$

$$s^{2} - \left(\frac{Y_{\beta}}{mV} + \frac{N_{r}}{J_{zz}}\right)s + \left(\frac{Y_{\beta}}{mV}\frac{N_{r}}{J_{zz}} - \frac{Y_{r}}{mV}\frac{N_{\beta}}{J_{zz}} + \frac{N_{\beta}}{J_{zz}}\right) = 0$$

This equation is similar to the one of single dof oscillating mass 

$$m s^{2} + c s + k = 0$$

$$s^{2} + \xi s + \Omega^{2} = 0$$

$$\xi = c/m \qquad \Omega^{2} = k/m$$

Х

Stability equation

$$s^2 + \xi \, s + \Omega^2 = 0$$

Roots of the characteristic equations

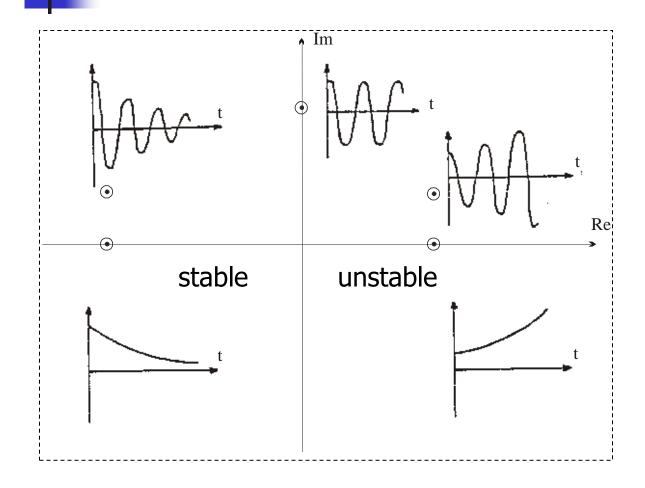
$$s_{1,2} = -\xi/2 \pm 1/2\sqrt{\xi^2 - 4\Omega^2}$$

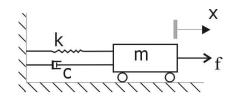
- Stability criterion: <u>The real parts of all roots must be negative</u>
  - In case of conjugate roots, their sum must be negative
  - In case of real roots, their sum must be negative and their product must be positive

That is:

$$s_1 + s_2 = -b/a < 0$$
  $s_1 \cdot s_2 = c/a > 0$ 

This criterion is equivalent to Routh Hurwitz.





$$m s2 + c s + k = 0$$
$$s2 + \xi s + \Omega2 = 0$$

Characteristic equations (reminder)

$$\begin{array}{c|c} mVJ_{zz} \ s^2 - \left(Y_\beta J_{zz} + mVN_r\right)s + \left(Y_\beta N_r - Y_r N_\beta + N_\beta mV\right) = 0 \\ \hline \mathbf{a} \qquad \mathbf{b} \qquad \mathbf{C} \end{array}$$

- To be checked  $s_1 + s_2 = -b/a < 0 \quad \Leftrightarrow \quad mVN_r + J_{zz}Y_\beta < 0$  $s_1 \cdot s_2 = c/a > 0 \quad \Leftrightarrow \quad mVN_\beta + Y_\beta N_r - Y_r N_\beta > 0$
- First condition: always satisfied

$$mVN_r = -m(b^2 C_{\alpha f} + c^2 C_{\alpha r}) < 0$$
$$J_{zz}Y_\beta = -J_{zz}(C_{\alpha f} + C_{\alpha r}) < 0$$

Second condition:

 $s_1 \cdot s_2 = c/a > 0 \quad \Leftrightarrow \quad mVN_\beta + Y_\beta N_r - Y_r N_\beta > 0$ 

• So  $mVN_{\beta} = -mV(b C_{\alpha f} - c C_{\alpha r})$   $N_rY_{\beta} = (C_{\alpha f} + C_{\alpha r})\frac{b^2 C_{\alpha f} + c^2 C_{\alpha r}}{V}$  $N_{\beta}Y_r = (b C_{\alpha f} - c C_{\alpha r})\frac{b C_{\alpha f} - c C_{\alpha r}}{V}$ 

$$N_r Y_\beta - N_\beta Y_r = \frac{C_{\alpha f} C_{\alpha r} L^2}{V}$$

It comes the condition

$$-mV(b C_{\alpha f} - c C_{\alpha r}) + \frac{C_{\alpha f}C_{\alpha r}L^2}{V} > 0$$

The second condition is satisfied if

$$-mV(b C_{\alpha f} - c C_{\alpha r}) + \frac{C_{\alpha f}C_{\alpha r}L^2}{V} > 0$$

- For an <u>understeer vehicle</u>:  $N_{\beta} = -(b C_{\alpha f} c C_{\alpha r}) > 0$ the dynamic behaviour is always stable
- For an <u>oversteer vehicle</u>  $N_{\beta} = -(b C_{\alpha f} c C_{\alpha r}) < 0$ the dynamic behaviour is <u>unstable</u> above the critical speed

$$V_{crit}^2 = \frac{C_{\alpha f} C_{\alpha r} L^2}{m(b C_{\alpha f} - c C_{\alpha r})} = -\frac{C_{\alpha f} C_{\alpha r} L^2}{m N_{\beta}}$$

#### Reminder: Neutral steer, understeer and oversteer vehicles

Understeer gradient in Steady State Cornering

$$K = \frac{m c}{C_{\alpha f} L} - \frac{m b}{C_{\alpha r} L} = m \frac{c C_{\alpha r} - b C_{\alpha f}}{C_{\alpha f} C_{\alpha r} L}$$

• If K=0, the vehicle is said to be *neutralsteer*:

 $K = 0 \quad \Leftrightarrow \quad c C_{\alpha r} = b C_{\alpha f} \qquad \qquad N_{\beta} = -(b C_{\alpha f} - c C_{\alpha r}) = 0$ 

• If K>0, the vehicle is *understeer* :

 $K > 0 \quad \Leftrightarrow \quad c C_{\alpha r} > b C_{\alpha f} \qquad \qquad N_{\beta} = -(b C_{\alpha f} - c C_{\alpha r}) > 0$ 

If K<0, the vehicle is *oversteer*:

$$K < 0 \quad \Leftrightarrow \quad c C_{\alpha r} < b C_{\alpha f} \qquad \qquad N_{\beta} = -(b C_{\alpha f} - c C_{\alpha r}) < 0$$

$$34$$

### Investigation of the motion nature

• Investigation of the discriminant of the stability equation  $\ddot{x} + \xi \dot{x} + \Omega^2 x = 0$ 

$$\rho = \xi^2 - 4\Omega^2$$

$$= \left(\frac{Y_\beta}{mV} + \frac{N_r}{J_{zz}}\right)^2 - 4\left(\frac{Y_\beta}{mV}\frac{N_r}{J_{zz}} - \frac{Y_r}{mV}\frac{N_\beta}{J_{zz}} + \frac{N_\beta}{J_{zz}}\right)$$

$$\xi = -\left(\frac{Y_\beta}{mV} + \frac{N_r}{J_{zz}}\right) \qquad \Omega^2 = \left(\frac{Y_\beta}{mV}\frac{N_r}{J_{zz}} - \frac{Y_r}{mV}\frac{N_\beta}{J_{zz}} + \frac{N_\beta}{J_{zz}}\right)$$

- If ρ>0: 2 real roots, damping is greater than the critical damping and one experiences an aperiodic motion.
- If ρ<0: 2 complex conjugate roots, damping is below the critical damping. One experiences an oscillation motion.</li>
- If ρ=0, 2 identical roots, critical damping of the system.

### Investigation of the motion nature

Value of the discriminant

$$\begin{split} \rho &= (\frac{Y_{\beta}}{mV} + \frac{N_r}{J_{zz}})^2 - 4\left(\frac{Y_{\beta}}{mV}\frac{N_r}{J_{zz}} - \frac{Y_r}{mV}\frac{N_{\beta}}{J_{zz}} + \frac{N_{\beta}}{J_{zz}}\right) \\ &= (\frac{Y_{\beta}}{mV} - \frac{N_r}{J_{zz}})^2 + 4\frac{Y_r}{mV}\frac{N_{\beta}}{J_{zz}} - 4\frac{N_{\beta}}{J_{zz}} \\ &= (\frac{C_{\alpha f} + C_{\alpha r}}{mV} - \frac{b^2 C_{\alpha f} + c^2 C_{\alpha r}}{J_{zz}V})^2 + 4\frac{N_{\beta}^2}{mV^2 J_{zz}} - 4\frac{N_{\beta}}{J_{zz}} \end{split}$$

• One finally finds

$$\rho = \left[ \left( \frac{C_{\alpha f} + C_{\alpha r}}{m} - \frac{b^2 C_{\alpha f} + c^2 C_{\alpha r}}{J_{zz}} \right)^2 + 4 \frac{(b C_{\alpha f} - c C_{\alpha r})^2}{m J_{zz}} \right] \frac{1}{V^2} - 4 \frac{N_\beta}{J_{zz}}$$

### Investigation of the motion nature

Discriminant (reminder)

$$\rho = \left[ \left( \frac{C_{\alpha f} + C_{\alpha r}}{m} - \frac{b^2 C_{\alpha f} + c^2 C_{\alpha r}}{J_{zz}} \right)^2 + 4 \frac{(b C_{\alpha f} - c C_{\alpha r})^2}{m J_{zz}} \right] \frac{1}{V^2} - 4 \frac{N_\beta}{J_{zz}}$$

- If  $N_{\beta} < 0$  (oversteer machine),  $\rho > 0$ .
  - The dynamic response is aperiodic
  - Stable as long as V < V<sub>crit</sub>.
- If  $N_{\beta} > 0$  (understeer machine),  $\rho < 0$ .
  - Positive term decreases as 1/V<sup>2</sup>
  - The response becomes oscillation (damped) over the speed

$$V_{osc.}^{2} = \frac{J}{N_{\beta}} \left[ \left( \frac{C_{\alpha f} + C_{\alpha r}}{m} - \frac{b^{2} C_{\alpha f} + c^{2} C_{\alpha r}}{J_{zz}} \right)^{2} + 4 \frac{(b C_{\alpha f} - c C_{\alpha r})^{2}}{m J_{zz}} \right]$$

### Steady state circular motion

#### Particular case: steady state turn

• The circular motion is characterized by:

$$\dot{\beta} = \dot{r} = 0$$
  $a_y = \frac{V^2}{R}$   $r = \dot{\theta} = \frac{V}{R}$ 

It comes

$$\begin{cases} -Y_{\beta} \beta + (mV - Y_r) r = Y_{\delta} \delta \\ -N_{\beta} \beta - N_r r = N_{\delta} \delta \end{cases}$$

- One extracts the value of the slip angle  $\beta = -\frac{N_{\delta} \ \delta + N_r \ r}{N_{\beta}}$
- The value of the yaw angle writes

$$(-mV N_{\beta} + N_{\beta}Y_r - N_rY_{\beta}) r = (N_{\delta}Y_{\beta} - N_{\beta}Y_{\delta}) \delta$$

#### Particular case: steady state turn

It yields the gain between the yaw speed and the steering angle:

$$\frac{r}{\delta} = \frac{N_{\beta}Y_{\delta} - N_{\delta}Y_{\beta}}{N_rY_{\beta} - N_{\beta}Y_r + mV N_{\beta}}$$

• Given that:  

$$N_{\beta}Y_{\delta} = C_{\alpha f}(cC_{\alpha r} - bC_{\alpha f})$$

$$N_{r}Y_{\beta} - N_{\beta}Y_{r} = \frac{C_{\alpha f}C_{\alpha r}L^{2}}{V}$$

$$\frac{-N_{\delta}Y_{\beta}}{N_{\beta}Y_{\delta} - N_{\delta}Y_{\beta}} = C_{\alpha f}(C_{\alpha r} + C_{\alpha f})$$

$$N_{\beta}Y_{\delta} - N_{\delta}Y_{\beta} = C_{\alpha f}C_{\alpha r}L$$

It comes

$$\frac{\delta}{r} = \frac{L}{V} + N_{\beta} \; \frac{mV}{C_{\alpha f} \; C_{\alpha r} \; L}$$

#### Particular case: steady state turn

• Now taking into consideration the circular motion nature  $r = \frac{V}{R}$ 

$$\delta = \frac{L r}{V} + N_{\beta} \frac{mVr}{C_{\alpha f} C_{\alpha r} L} \qquad \qquad \delta = \frac{L}{R} + \frac{m N_{\beta}}{C_{\alpha f} C_{\alpha r} L} \frac{V^2}{R}$$

And by introducing the value

$$N_{\beta} = -(b C_{\alpha f} - c C_{\alpha r})$$

One recovers the classical expression for steady state turn

$$\delta = \frac{L}{R} + \left(m\frac{c}{L}\frac{1}{C_{\alpha f}} - m\frac{b}{L}\frac{1}{C_{\alpha r}}\right)\frac{V^2}{R}$$

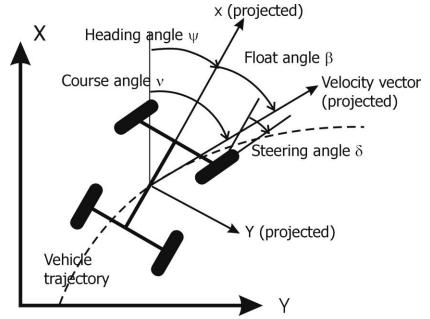
### Trajectory description

### **Trajectory description**

The vehicle trajectory can be described using a parametric equation relating the space coordinates and the time

 $t\mapsto (\bar{X}(t),\bar{Y}(t))$ 

- One defines:
  - θ the course angle between
     the trajectory and the
     frames axis X
  - ψ the heading angle between the X of the reference frame and the x axes of the car
  - β the side slide angle
     of the vehicle, the angle between the vehicle axis x and the velocity vector tangent to the trajectory



### **Trajectory description**

• We can write the following relations between the course angle  $\theta$ , the heading angle  $\psi$  and the side slip angle  $\beta$ :

$$\vartheta = \psi + \beta \qquad \qquad \dot{\vartheta} = r + \dot{\beta}$$

 The linear velocities are obtained by using frame transformation between the vehicle body axes and the inertial reference frame

$$\frac{dX}{dt} = V\cos\vartheta \qquad \qquad \frac{dX}{dt} = u\cos\psi - v\sin\psi$$
$$\frac{dY}{dt} = V\sin\vartheta \qquad \qquad \frac{dY}{dt} = u\sin\psi + v\cos\psi$$